MATERIAL CHARACTERIZATION OF A NORMALIZED AND TEMPERED, 0.2 w/o C CAST STEEL

by

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ABSTRACT

Monotonic and cyclic stress-strain behavior and strain-life resistance of a 0.2 w/o C cast steel of 160 BHN are reported. This cast steel exhibits a monotonic 0.2% yield strength of 49 ksi and a minimum cyclic 0.2% yield strength of 47 ksi. The upper yield point is 58.5 ksi. However, due to cyclic softening, the cyclic flow stress is reduced to approximately 25 ksi. Engineering ultimate strength and ductility are comparable to a similar wrought steel of equal hardness.

As with other cast metals in which internal microdiscontinuities are present, the strain-life fatigue resistance is comparable to but slightly inferior to that of a wrought steel of equal hardness. Recently developed predictive techniques, which employ measurements of the size and distribution of the largest microdiscontinuities and matrix hardness measurement, produce an accurate description of the fatigue resistance of this cast steel.

A Report of the FRACTURE CONTROL PROGRAM

College of Engineering, University of Illinois Urbana, Illinois 61801 October, 1975

ACKNOWLEDGMENTS

Technical discussions, criticisms and manuscript reviews of Professors

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FOREWORD

This is the third in a series of the fatigue evaluation of steels of interest to sponsors of the Fracture Control Program (see FCP Nos. 12 on T-1 Steel and 13 on Cast 8630 Steel). General Motors Corporation, Electromotive Division supplied a cast, 0.2 w/o C steel at 160 BHN, in normalized and tempered condition for evaluation in this report. As in previous reports of this nature, the format includes reduced material characterization sheets as well as original laboratory records.

PROCEDURE

Specimens of the design shown in Fig. 1 were removed from cast keel blocks which were received in a normalized $(1700^{\circ} \, \text{F})$ and tempered $(1200^{\circ} \, \text{F})$ condition with a Brinell hardness of 160. All testing was accomplished using a ± 20 kip closed-loop test system.

RESULTS

Stress-Strain Behavior

On the Data Sheet for Material Characterization are listed the results from the monotonic stress-strain tests. Since these cast samples are of moderately low hardness with approximately a 75% ferritic, 25% pearlitic matrix and the internal microdiscontinuities are not significantly large, as shown in Fig. 2, the observed monotonic strength and ductility are comparable to a wrought steel of equal hardness.

As shown in Fig. 3, which is an enlargement of the stress-strain curves on the data sheets, the cast steel cyclically softens at strains less than $\pm 0.0046^*$ and cyclically hardens at larger strains. Note also that there is a significant decrease

^{*}From companion specimen cyclic stress-strain curve.

in the initial monotonic and cyclic "flow" strength (i.e., 59 ksi upper yield strength compared to 25 ksi). Consequently, in the stress-based design of a component to a maximum stress of 50 ksi (i.e., the lower yield strength) an "elastic" strain of approximately 0.0017 would be produced on the first application of load. Due to cyclic softening, however, a significant amount of plastic strain ($\epsilon_p \cong 0.0005$) would eventually result, since the material response follows the cyclic stress-strain curve. This would eventually cause a fatigue failure in about 10^5 reversals.* Secondly, the cyclic curve shown in the figure is essentially for a "bulk" or nominal stress-strain response of the cast steel. Since there are casting microdiscontinuities present, the <u>local</u> stresses and strains at these concentrations would be even greater than shown.

As demonstrated in FCP Report No. 13 on cast 8630 steel, employing the monotonic and cyclic properties to predict the strain-life response of cast metals can lead to serious errors in lives. It is somewhat erroneous to even list cyclic material properties for cast metals, since <u>local</u> stresses and strains at casting microdiscontinuities govern behavior. These properties are included on the data sheet for convenience only and should be used with caution.

Strain-Life Resistance

1

Figure 4 illustrates the constant amplitude, strain-life curves for single samples which received an initial precycle of 5 cycles at ±0.013 with an incremental decrease of amplitude to zero stress and strain in 20 cycles. These results are listed at the back of the report.

Superimposed in Fig. 5 is a strain-life curve predicted from hardness results (see Appendix A) and considering the cast steel to be wrought. As shown, such attempts may result in non-conservative predictions by orders of magnitude in life,

^{*}See Fig. 5. At a strain of 0.0017, $2N_f^{\ \simeq\ 10^5}$ reversals.

because the effect of microdiscontinuities is not taken into account. Figure 6 shows micrographs of the fracture surface of the fatigue specimens. Note that all failures initiated from gas bubbles at or in the proximity of surfaces.

Predicted Strain-Life Curve for Cast Steel

It has been suggested in previous FCP reports (14 and 15) that the size, shape and distribution of microdiscontinuities must be included in life predictions for cast irons.

Since the largest discontinuities, particularly those at or in proximity to the surface of a sample, contro! fatigue behavior, they are the most important.* Table 1 lists results of microdiscontinuity diameter measurements made on four specimens whose gage length (i.e., the stressed volume) was divided into eight sectors. Only the largest discontinuity visible at 7X in each sector is listed. If no "large discontinuity" is observed, a dash is recorded but these "no show" observations are included in the subsequent ranking of these data which is shown in Table 2. These results, plotted as log-normal and Weibull distributions, are shown in Figs. 7 and 8.

To determine a fatigue notch factor, K_f , for the microdiscontinuities, the recently proposed procedure outlined in FCP No. 15 was employed. Using an average diameter (d $\stackrel{\checkmark}{=}$ 0.016 in) of largest microdiscontinuities from the four samples (i.e., 0.02047, 0.01968, 0.01181 and 0.00984), a value for the fatigue notch factor, K_f , was determined by

$$K_f = 1 + \frac{K_t - 1}{1 + a/r}$$

Shape $\begin{cases} K_t = \text{ theoretical stress concentration factor for hemispherical shaped} \\ \text{surface discontinuity (} \stackrel{\triangle}{=} 2.5) \end{cases}$

^{*}This, of course, presumes a uniform, random distribution of largest flaws or "weakest links."

a =
$$\frac{300}{0.5 \text{ BHN}}$$
 1.8 x 10⁻³ in = 0.011 in
Size $\left\{ \mathbf{r} = \frac{0.016}{2} = 0.008 \text{ in} \right.$
 $\therefore K_f = 1.6$

Incorporating results of the wrought steel prediction shown in Fig. 5, values for the Neuber parameter, $(\Delta\sigma \ \Delta\epsilon \ E_s)^{\frac{1}{2}}$, were obtained. This particular parameter would describe the fatigue resistance of the equivalent wrought steel. The quotient $(\Delta\sigma \ \Delta\epsilon \ E_s)/K_f = (\Delta S \ \Delta\epsilon \ E_c)^{\frac{1}{2}}$ is plotted in Fig. 9 in comparison to the Neuber parameter from the cast steel test results. Since the average diameter of the largest microdiscontinuities was used to determine K_f , this curve is for median behavior. Therefore, employing the size and shape of the average largest surface microdiscontinuity to determine a fatigue notch factor and appropriately modifying the Neuber parameter for the matrix steel provides a predicted cast steel curve. As shown, the agreement is quite good.

In the predictive technique outlined above, there have been several assumptions, namely that initiation of a fatigue crack occurs at surface imperfections. The rationale of this assumption is the increase in value of the theoretical stress concentration factor, K_t , as a microdiscontinuities position in a sample varies from center to surface. The value of K_t approaches 2.0 for a centerline spherical discontinuity and increases as the imperfection approaches the sample surface. Secondly, surface residual stresses are not, as yet, taken into account. If, for example, a sizable compressive residual stress exists, there is the possibility of sub-surface initiation as in the case of shot-peened or case hardened components. Thirdly, only a relatively small sample volume was scanned in order to ascertain the average "largest" microdiscontinuities. In order that this analysis be extended to larger volumes, a function describing the distribution of microdiscontinuities must be employed.

An approximate measure of the volume scanned during these observations, shown in Fig. 10, is given by

$$V_0 = \pi [r_0^2 - (r_0 - D)^2] \ell \cdot n$$

where; r_0 = radius of gage section of specimen (0.1375 in)

D = average diameter of largest surface microdiscontinuities (D = \overline{X} = 0.011 in)

le = gage length of specimen (0.75 in)

n = number of specimens (4)

...
$$V_0 = 0.019 \text{ in}^3$$

In fatigue, the ratio V/V_0 may be treated as the number of "links" in a chain. If the probability of survival (one minus the probability of failure) is $1 - p_0$ when the sample size is V_0 , then the probability of survival for a volume V, will be

$$(1 - p) = (1 - p_0)^{V/V_0}$$

As an example, assume the population of microdiscontinuities is described by the log-normal distribution shown in Fig. 7. The median flaw diameter (at $p_0 = 0.5$) is 0.0036 in. For purposes of illustration, assume that a component with 100 times the volume of highly stressed material is to be considered. This corresponds to the p = 0.999 level for the smaller sample. The microdiscontinuity diameter at that level is 0.07 in. For a surface imperfection of this size, the value of K_f is approximately 2.2, which was used in the construction of the "lower bound" fatigue curve shown in Fig. 9. In such an extrapolation it is presumed that the foundry practice and cooling rate of the larger castings is equivalent to those of the kccl blocks from which the specimens were removed.

Summary of Predictive Technique for Fatigue Behavior of Cast Irons

From surface observations of four specimens, the average largest micro-discontinuity was determined. Using a matrix hardness measurement, the average largest microdiscontinuity size, and a theoretical stress concentration factor a value of the fatigue notch factor was determined. The "average" fatigue resistance for this cast steel was then predicted. The good agreement between predicted and actual results appears to be excellent, indicating that further research merits attention. In order to extrapolate to the maximum size microdiscontinuities expected in a larger volume of material, a distribution function was introduced.

Conclusions

Monotonic stress-strain results of cast 0.2 w/o C steel at 160 BHN are consistent with a similar wrought steel. Cyclic stress-strain results indicate significant softening at small strains. Methods which employ matrix hardness and the size and shape of microdiscontinuities adequately predict fatigue resistance. A lower bound curve of fatigue resistance in larger volumes of material may be established by introducing a distribution function for the largest microdiscontinuities observed in the sample volume.

TABLE 1
LARGEST SURFACE DISCONTINUITY OBSERVED AT 7X, (INCHES)*

				Sector No.	No.			
Sample No.	A1	A2	A3	A4	B1	B2	В3	B4
J3-12	0.00984	0,00669	ı	ı	ı	0.00708	ı	1
13-9	0.02047	1	1	•	1	1		ı
J3-3	ı	1	ı	0.01968	1	i	τ	ı
13-6	0.00393	ι	0.01181	0.01023	0.00787	ı	0,01181	ı

Average diameter $\overline{X} = 0.011$ inches.

*Converted from millimeters.

- Indicates "no show" observation of discontinuity at 7X.

TABLE 2

MEDIAN RANKING OF 32 OBSERVATIONS

Order	Dia. (inches)	Rank
23	0.00393	0.7006
24	0.00669	0. 7006 0. 7315
25	0.00708	0. 7623
26	0.00787	0.7932
27	0.00984	0.8240
28	0.01023	0.8549
29	0.01181	0.8858
30	0.01181	0.9166
31	0.01968	0.9474
32	0.02047	0. 9783

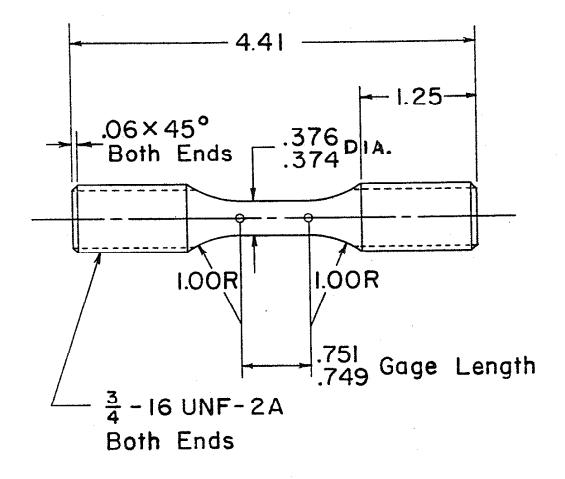
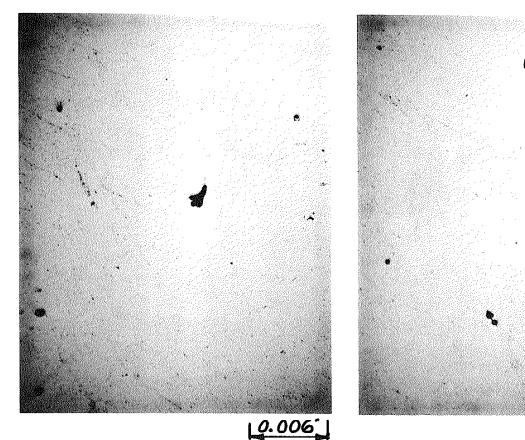
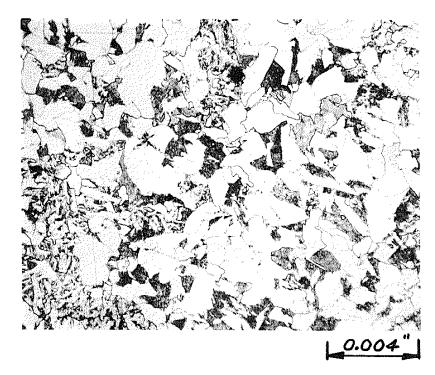


FIG. I - SPECIMEN DESIGN



TYPICAL MICRODISCONTINUITIES - UNETCHED



FERRITIC-PEARLITIC MATRIX - 2% NITAL
FIGURE Z - MACROGRAPHS SHOWING TYPE I INCLUSIONS & GAS BUBBLES
AND MICROGRAPH OF 0.2% C CAST STEEL

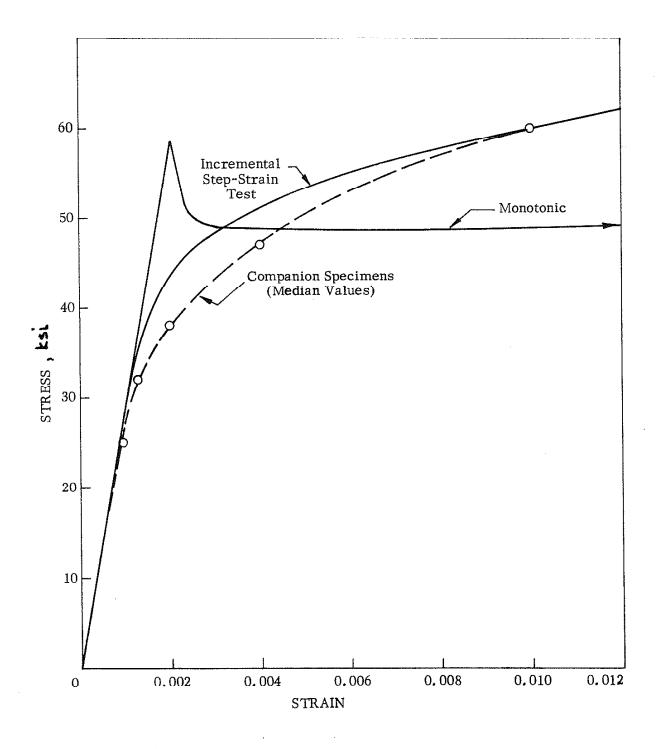


FIGURE 3 - MONOTONIC & CYCLIC STRESS - STRAIN

CURVES FOR D.Z%C CAST STEEL @160 BHN

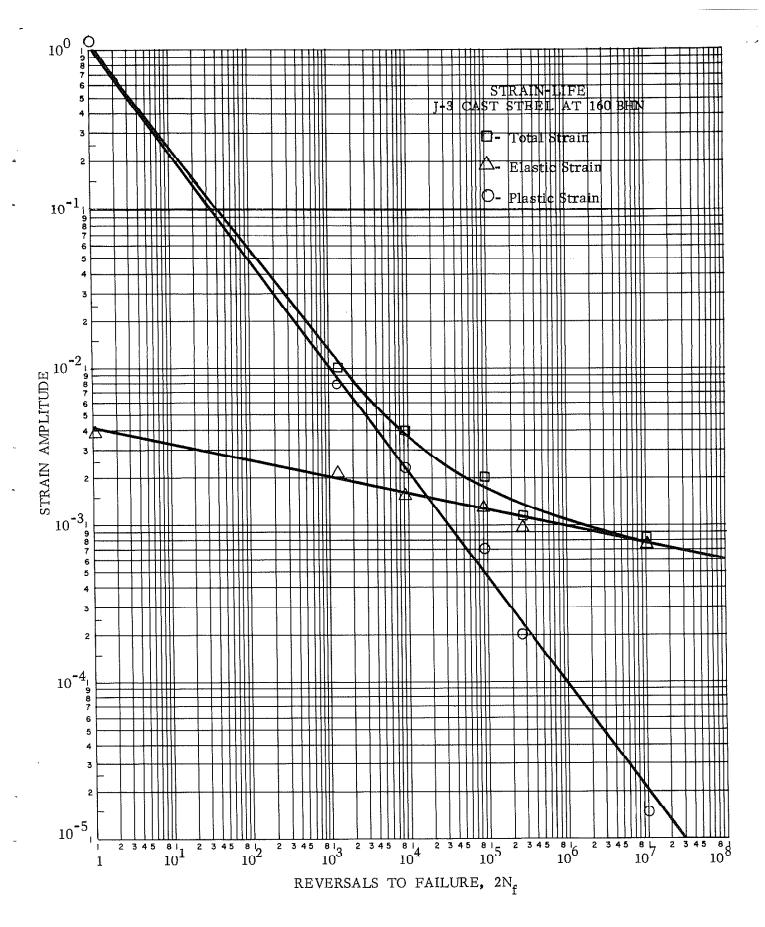


FIGURE 4 - STRAIN-LIFE CURVE FOR CAST 0.Z %C

Approximation of the state of t

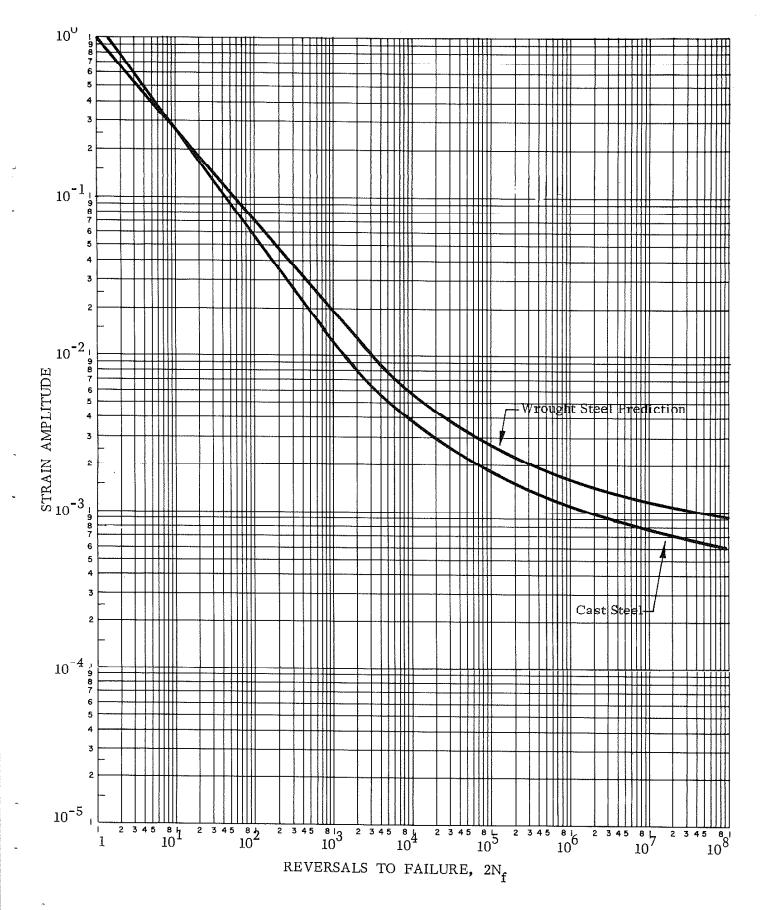
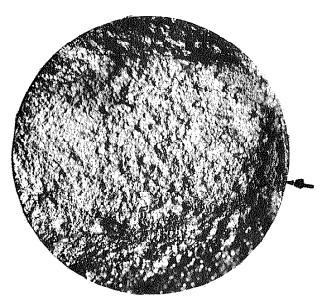


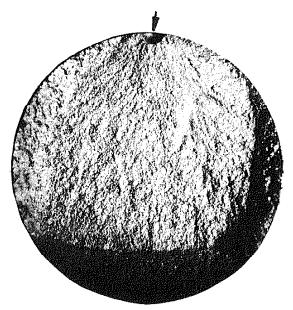
FIGURE 5 - STRAIN-LIFE FOR CAST 0.2% C STEEL IN

COMPARISON TO AN EQUAL HARDNESS WROUGHT

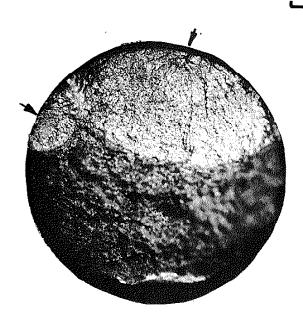
STEEL (IGOBHN).



10/2 = 0.01 2N_p = 1,300 revs

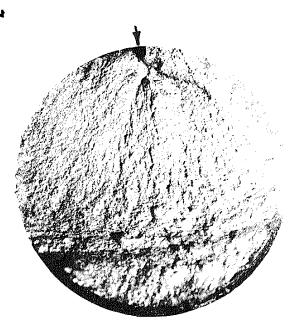


Δe12 = 0.004 2N_f = 9,200 revs



10/2 = 0.002 2Nf = 9.1 X104 revis

- = INITIATION SITE



10.0013 2Ng= 2.7x 10 revis

FIGURE 6 - MACROGRAPHS OF FRACTURE SURFACES
OF FATIGUE SPECIMENS.

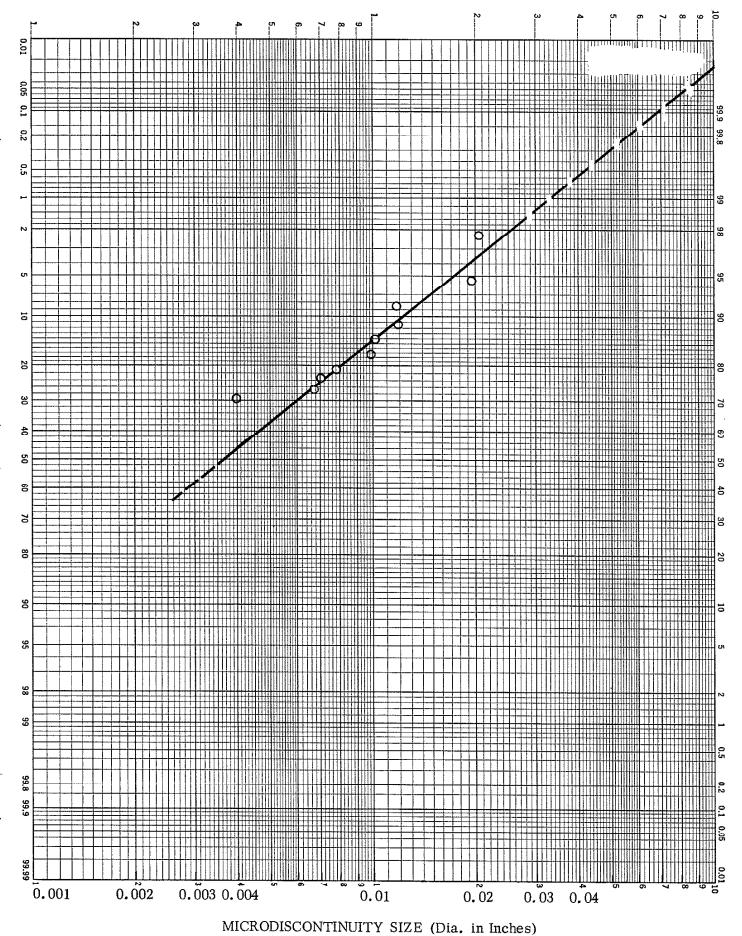


FIGURE 7 - LOG-NORMAL PLOT OF LARGEST MICRODISCONTINUITIES

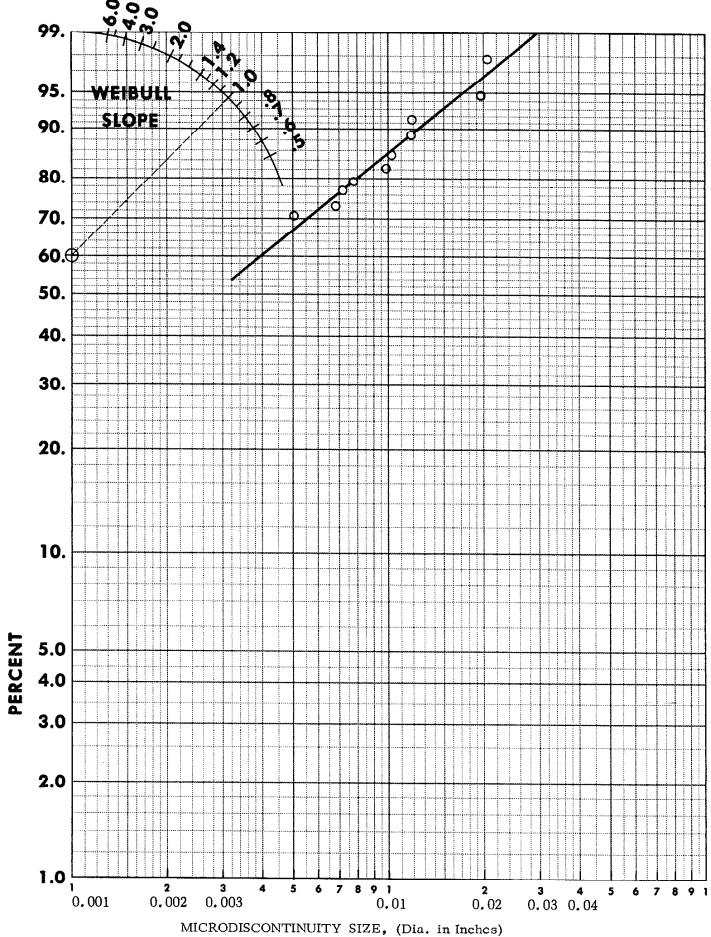


FIGURE 8 - WEIBULL PLOT OF LARGEST MICRODISCONTINUITIES

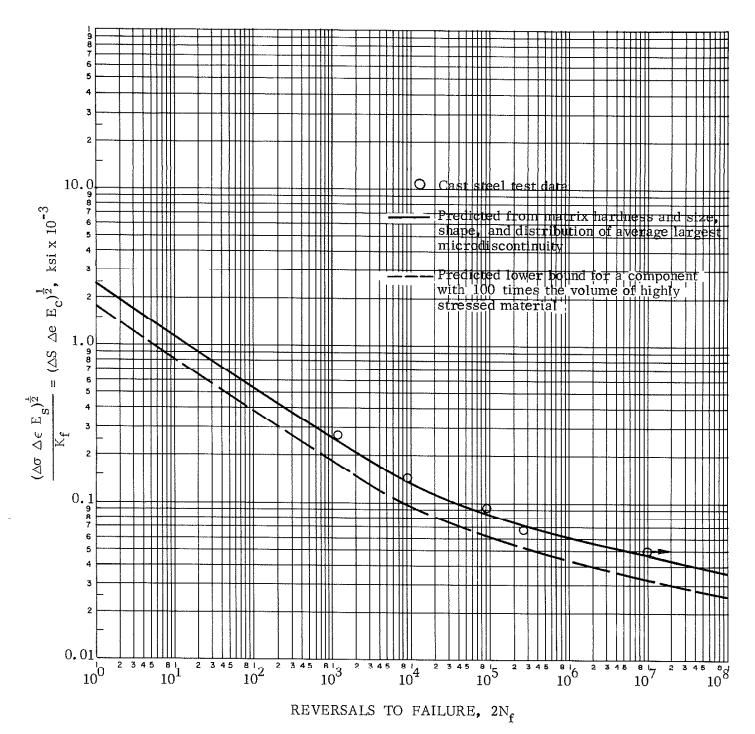


FIGURE & - PREDICTED STRAIN·LIFE CURVES FOR CAST 0.2%C STEEL
@160 BHN.

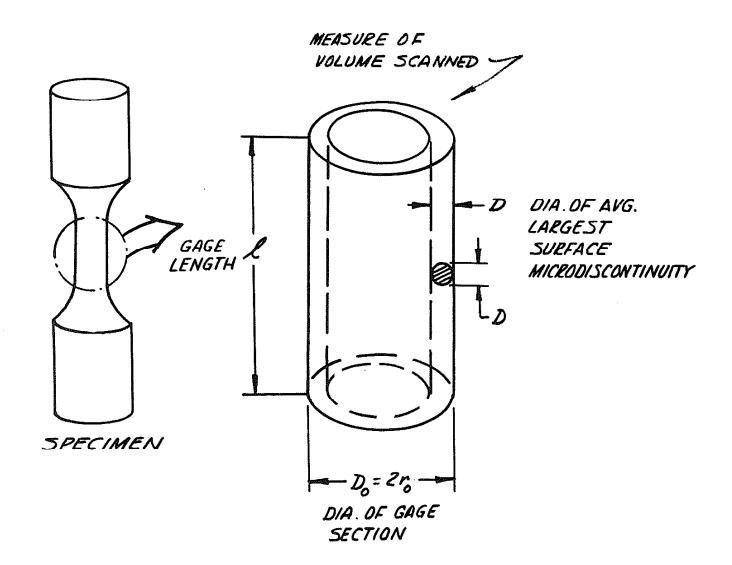


FIGURE 10 - SCHEMATIC OF VOLUME SCANNED

DURING MACROGRAPHIC OBSERVATIONS.

APPENDIX A

STRAIN-LIFE FATIGUE PREDICTION FOR A WROUGHT STEEL OF COMPARABLE HARDNESS (BHN = 160)

$$S_{ult}$$
 (ksi) $\stackrel{\ \ \! \square}{=} 0.5$ BHN $\stackrel{\ \ \! \square}{=} 0.5$ (160) = 80 ksi
$$\sigma_{f}' = \sigma_{f} \stackrel{\ \ \! \square}{=} 50 + S_{u}$$
 (ksi) = 50 + 80 = 130 ksi
$$E = 30 \times 10^{3} \text{ ksi}$$

$$\therefore \frac{\sigma_{f}}{E} = \frac{130}{30 \times 10^{3}} = 0.00433$$

$$b \stackrel{\ \ \! \! \square}{=} -\frac{1}{6} \log \frac{2\sigma_{f}}{S_{ult}} \stackrel{\ \ \! \! \square}{=} -0.085$$

Let $\epsilon_f' = \epsilon_f \stackrel{\text{def}}{=} 1$ since steel is of "lower" hardness and let c = -0.6

MATERIAL CHARACTERIZATION SHEETS

DATA SHEET FOR MATERIAL CHARACTERIZATION

Material: J-3 Cast Steel	Matrix Hardness:	160 BHN
Condition: Normalized at 1700°F Tempered at 1200°F	Converted from:	RB
Monotonic Properties:	Cyclic Properties:	
Modulus of Elasticity, E 28.6 x 10 ³ ksi	Yield Strength, 0.2% S	47.0* ksi
Yield Strength, 0.2% S _v 49.0 ksi	Strain Hardening Exponent, n'	0.12
Ultimate Strength, S _u 75.0 ksi	Strength Coefficient, K'	107 ksi
Red. in Area, % RA 67.8%	Fatigue Strength Coefficient, $\sigma_{ extsf{f}}'$	r ksi
True Fracture Strength, $\sigma_{\rm f}$	Fatigue Ductility Coefficient, $\epsilon_{ m f}^{'}$ _	1
True Fracture Ductility, $\epsilon_{\rm f}$ 113.0 ksi (corrected) 1.13	Fatigue Strength Exponent, b	-0.11
Strain Hardening Exponent, n 0.14	Fatigue Ductility Exponent, c	-0.68
Strength Coefficient, K	Transition Fatigue Life, 2N _t	19,000 rev
True Touchness. [] ksi	*From companion specimen test results	ılts

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Composition:

w/o C

Magnification:

0.16 0, 14 11

w/o Mo w/o Cu w/o Ni w/o Va w/o Al

Comments:

0,035 0.001

H 11

= 1.13= 0.13

w/o Mn w/o Cr

11

= 0.015= 0.010

w/o P S o/w

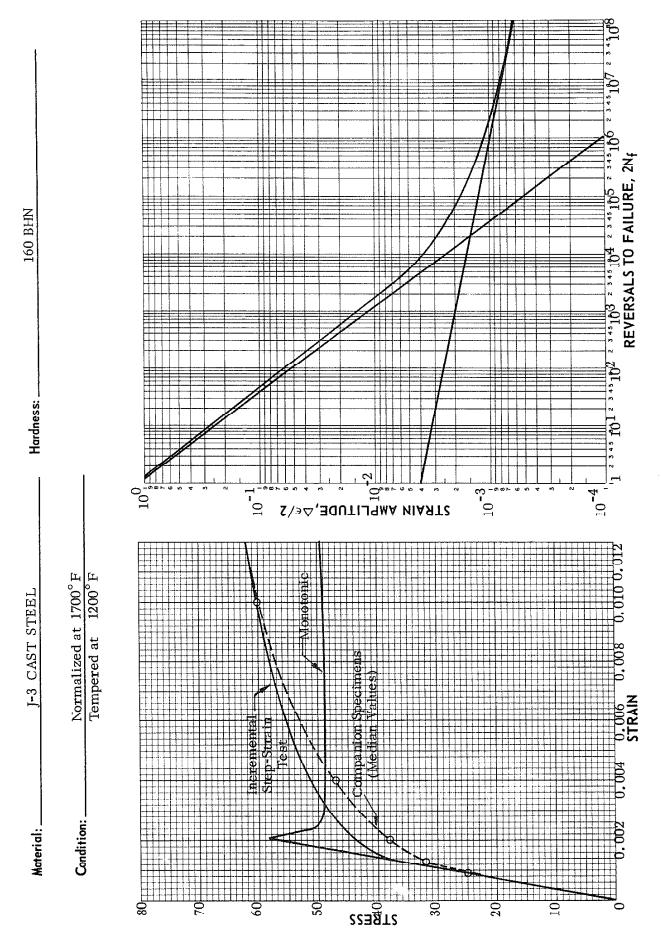
= 0.44= 0.20

w/o Si

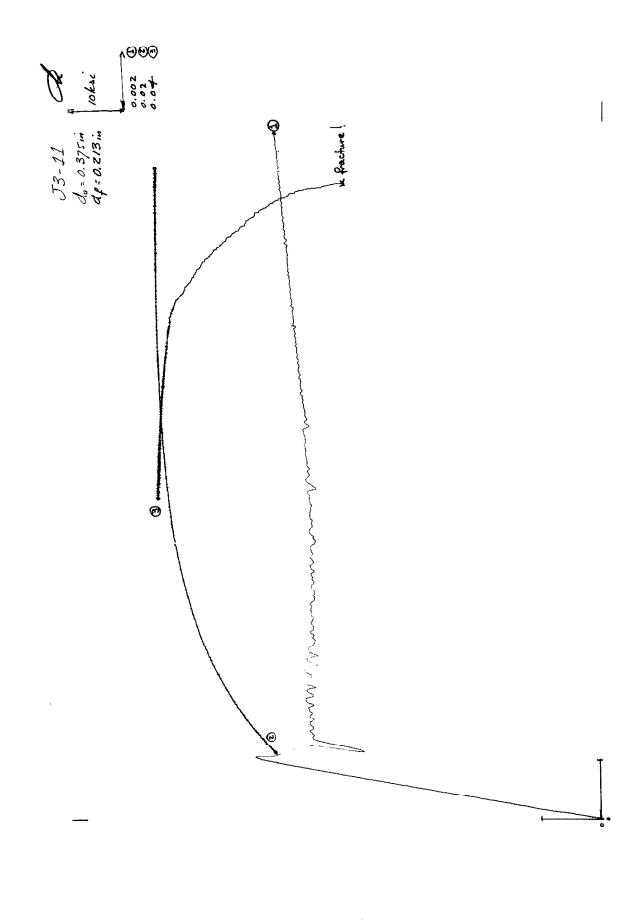
Grain Size:

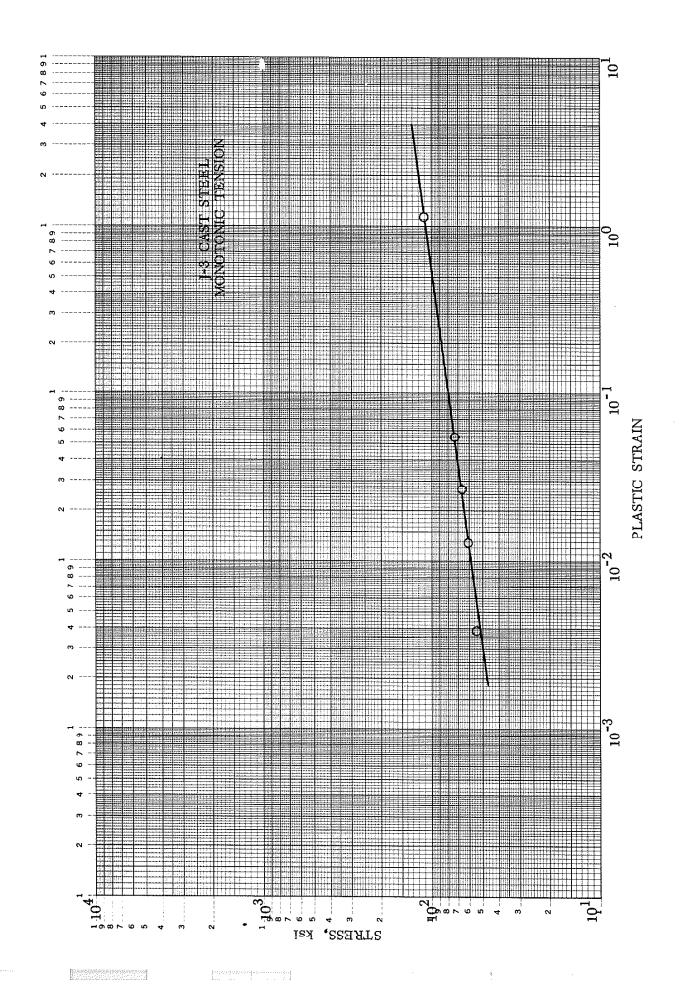
Eutectic Cell Size (Cast irons):

FRACTURE CONTROL PROGRAM UNIVERSITY OF ILLINOIS



MONOTONIC STRESS-STRAIN RESULTS





CYCLIC STRESS-STRAIN RESULTS

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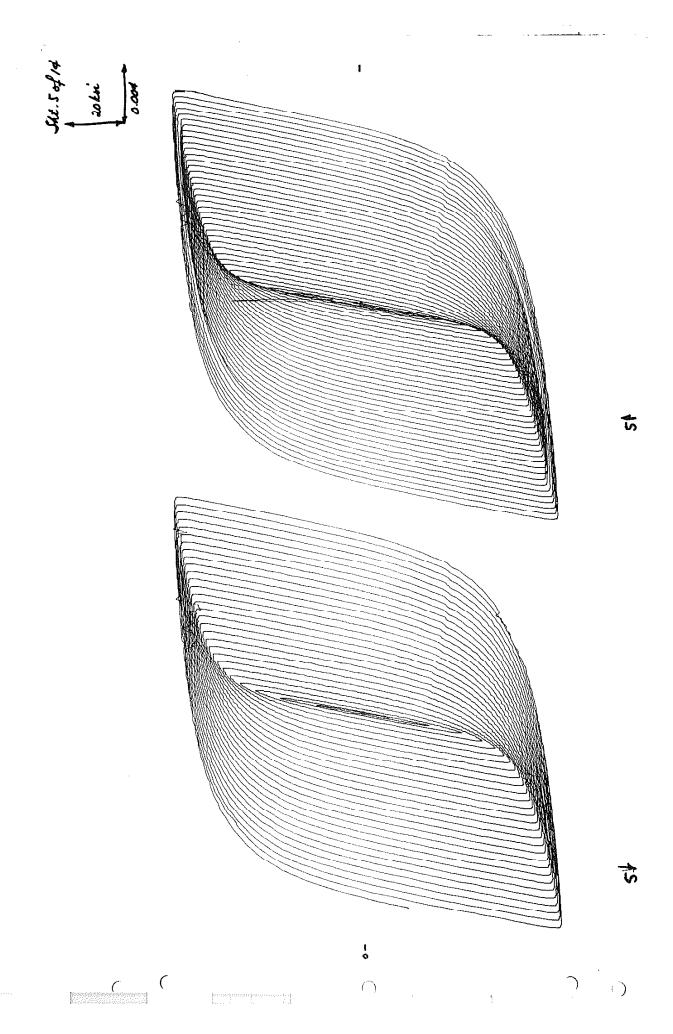
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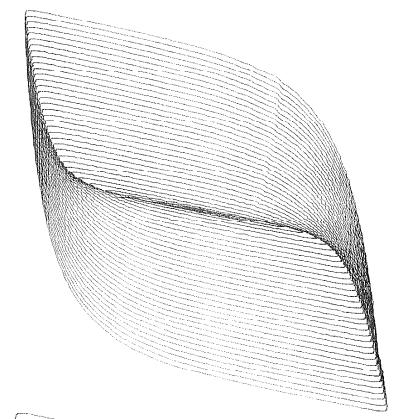
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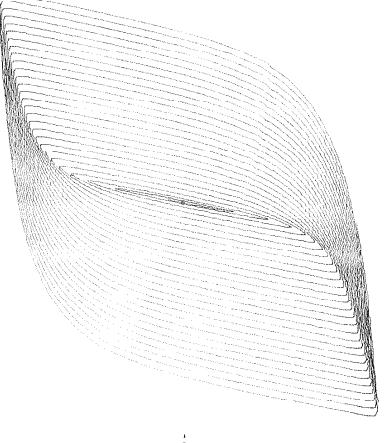
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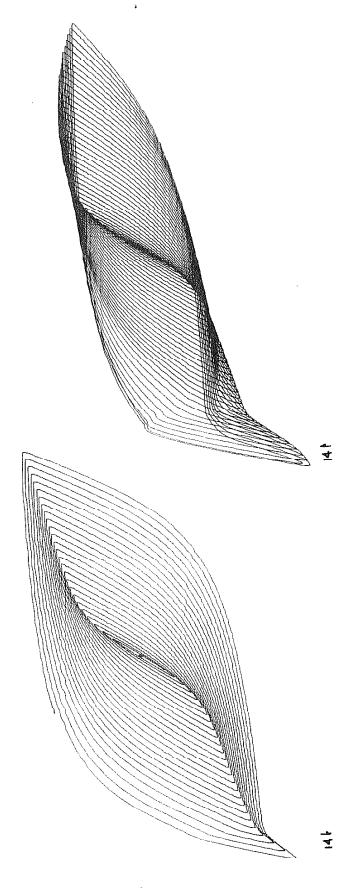
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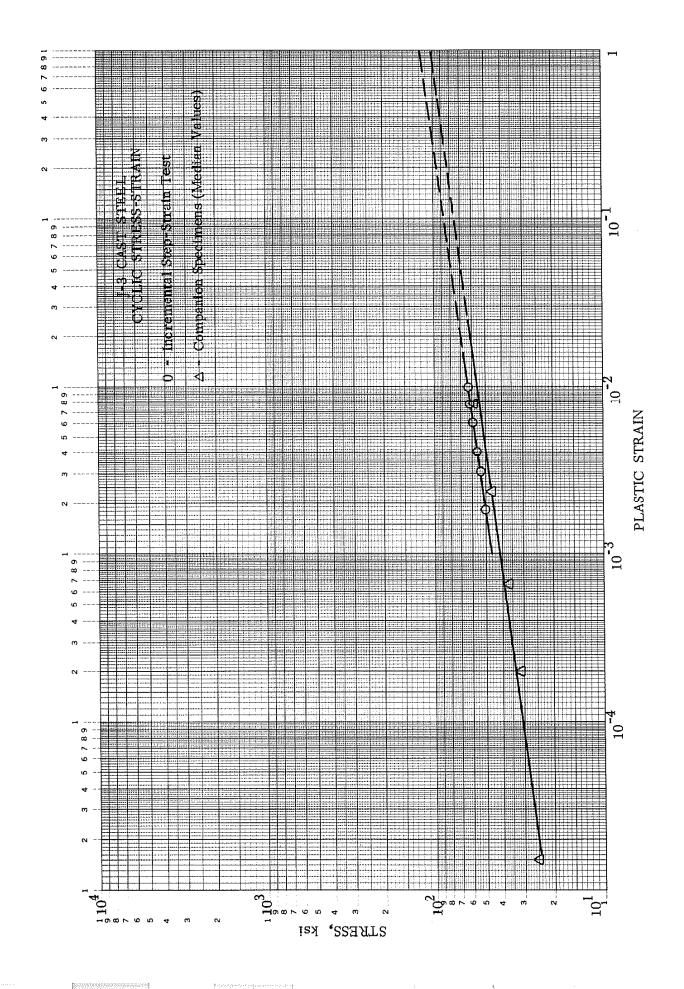


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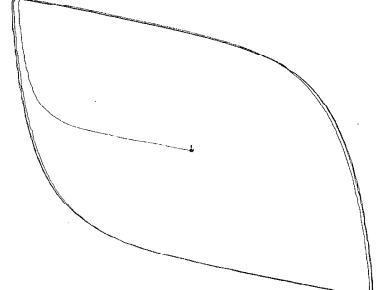


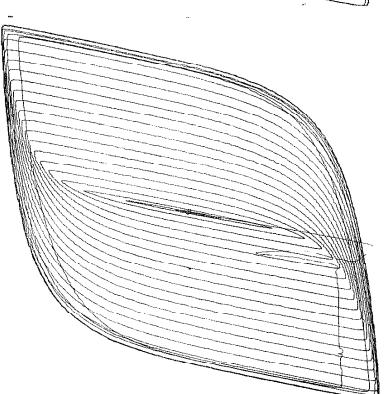




STRAIN-LIFE RESULTS

33-8 41143 do:0.576. 2004 Ap= 650~ 2004





Precycle 5 cyc @ 20.013
20 cyc dec. to 0.0.

0 20~ 25% 10~ •

She 20 f 3

J3-8

423.43 20kn

13-8

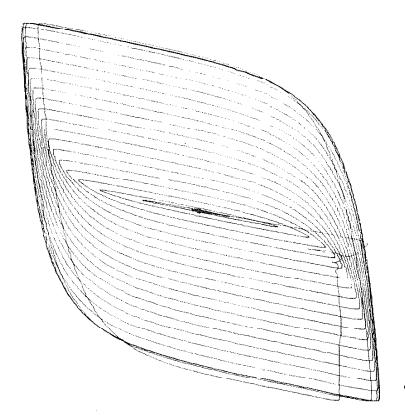
23-7

4=0.39

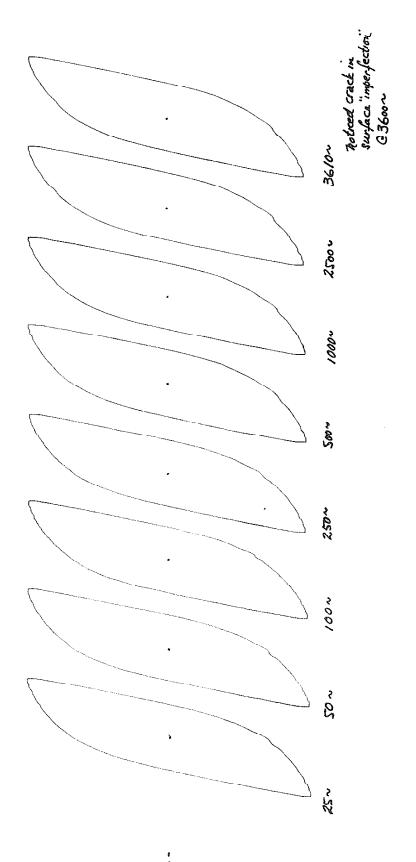
200- 2004

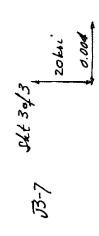
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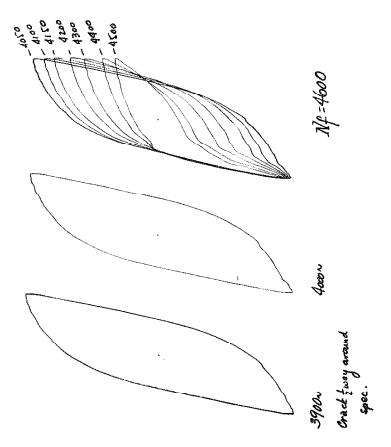
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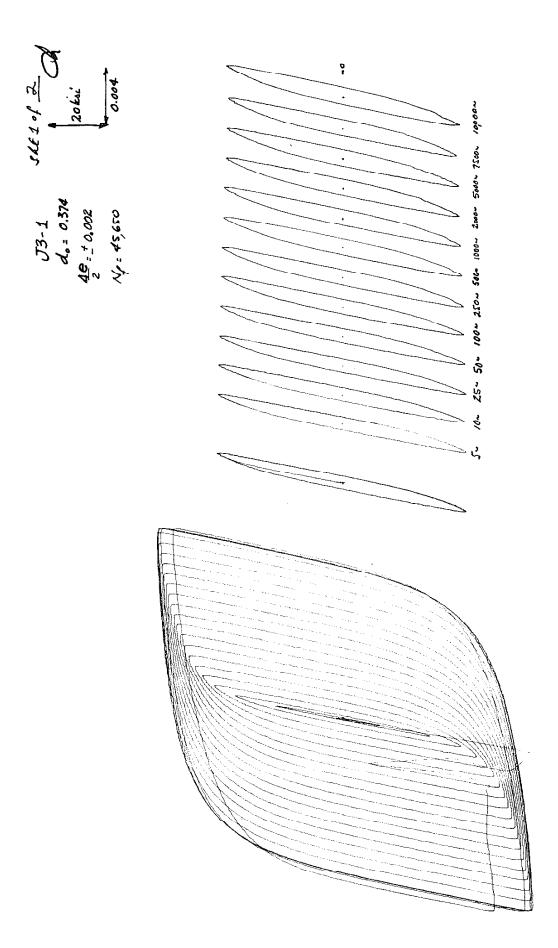


Precycle Scycles @ 10.013 Locye, deprease to 0-0.









J3-1

Np: 45650 15,000 m 13,000~

D.004 Sht 1 of 4

 $\sqrt{3}-2$ $d_0 = 0.376$ $\frac{\Delta^{\epsilon}}{2} = \pm 0.00/3$ $M_p = 733,750$

for ~ 500 - 100 . 5000 . 82 gas walking . . respect

33-4 Milsofs

do = 0.37/

ds = 2.37/

As= ± 25 km (0.000845e)

TRESS CONTROL

N=5.04 × 10⁶ runout!

STRAIN-LIFE RESULTS

pecimen Number	Strain Amplitude	Revs. to Failure 2N _f	Elastic Strain Amplitude $\Delta\epsilon_{\mathbf{e}}/2$	Plastic Strain Amplitude △ _{←p} /2	Saturation* Stress Amplitude △o/2, ksi
J3-8	0.01	1,300	0.0021	0.0079	60.0
J3 - 7	0.004	9,200	0.00164	0.00236	47.0
J3-1	0.002	9.1×10^4	0.00133	0.00067	38.0
J3-2	0.0013	2.7×10^5	0.00110	0.0002	32.0
J3 - 4	0.00089	$1.0 \times 10^{7**}$	0.000875	0.000015	25.0
J3-7 J3-1 J3-2	0.004 0.002 0.0013	9,200 9. 1×10^4 2. 7×10^5	0.00164 0.00133 0.00110	0.00236 0.00067 0.0002	

Note: All specimens received an initial precycle of 5 cycles at ± 0.013 which was then incrementally decreased to zero stress and strain in 20 cycles.

^{*}Measured at 50% of life to failure.

^{**}Runout, specimen did not fail!

STRESS-TIME RESPONSE TO CONTROLLED STRAIN FATIGUE TESTS

ტ : 4 = დ : SEMILOGARITMBIC 7 CYCLES X CO FLICTONS REUFFRE & RESER CO. Ž W

DATE

MODEL

ADDITIONAL MATERIAL CHARACTERIZATION SHEE'IS

DATA SHEET FOR MATERIAL CHARACTERIZATION

4

Material:	Material: J-3 Cast Steel	Matrix Hardness:	160 BHN
Condition:	Condition: Normalized at 1700°F	Converted from:	
	Tempered at 1200°F		
Monotonic	Monotonic Properties:	Cyclic Properties:	

Modulus of Elasticity, E	$28.6 \times 10^3 \text{ ksi}$	Yield Strength, 0.2% Sy
Yield Strength, 0.2% S _v	49.0 ksi	Strain Hardening Exponent, n'
Ultimate Strength, S _u	75,0 ksi	Strength Coefficient, K'
Red. in Area, % RA	67.8%	Fatigue Strength Coefficient, $\sigma_{ extsf{f}}^{ extsf{'}}$
True Fracture Strength, of	134, 0 ksi	Fatigue Ductility Coefficient, $\epsilon_{ m f}$
True Fracture Ductility, $\epsilon_{ m f}$	1.13	Fatigue Strength Exponent, b
Strain Hardening Exponent, n	0.14	Fatigue Ductility Exponent, c
Strength Coefficient, K	110 ksi	Transition Fatigue Life, 2N _t _
True Toughness, U.	112 ksi	*From companion specimen test

47.0* ksi	n' 0.12	107 ksi	t, of k	it, ef'	b - 0.11	c -0.68	t 19,000 rev	test results
Yield Strength, 0.2% S.'	Strain Hardening Exponent, n'	Strength Coefficient, K'	Fatigue Strength Coefficient, $\sigma_{ extbf{f}}$	Fatigue Ductility Coefficient, $\epsilon_{ m f}^{'}$	Fatigue Strength Exponent, b	Fatigue Ductility Exponent, c	Transition Fatigue Life, $2N_{ m t}$	$^* {\it From}$ companion specimen test results

Microstructure:	Magnification:		Comments:				
	= 0.19	= 0,16	= 0.14	1	= 0.035	= 0.001	
	w/o Mo	w/o Cu	Z	>	A (w/o B	
	0/M	M/C	w/o Ni	w/o Va	w/o Al	w/c	
	= 0.20	= 0.44	= 0.015	= 0.010	= 1,13	= 0.13	
ion:	Ŋ	Si	Д,	S	Mn	Cr	
Composition:	w/o C	w/o Si	w/o P	w/o S	w/o Mn	w/o Cr	

True Toughness, Up

Grain Size:

